

Implementation of the Johnson-Holmquist II (JH-2) Constitutive Model Into DYNA3D

by George A. Gazonas

ARL-TR-2699 March 2002

20020528 065

Approved for public release; distribution is unlimited.

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

Citation of manufacturer's or trade names does not constitute an official endorsement or approval of the use thereof.

Destroy this report when it is no longer needed. Do not return it to the originator.

Army Research Laboratory

Aberdeen Proving Ground, MD 21005-5069

ARL-TR-2699 March 2002

Implementation of the Johnson-Holmquist II (JH-2) Constitutive Model Into DYNA3D

George A. Gazonas Weapons and Materials Research Directorate, ARL

Approved for public release; distribution is unlimited.

Abstract

This report describes the implementation of a fully three-dimensional rate, pressure, and damage-dependent constitutive model for brittle materials such as ceramics into the explicit, Lagrangian finite element code DYNA3D. The model, otherwise known as the Johnson-Holmquist II (JH-2) ceramic model, has also been implemented into CTH, EPIC, and LS-DYNA, and is used extensively in modelling the brittle response of ceramics in armor applications. The DYNA3D material driver was used to verify the model implementation for constant strainrate input histories (Johnson, G. R., and T. J. Holmquist. "An Improved Computational Constitutive Model for Brittle Materials." High Pressure Science and Technology, New York: AIP Press, 1993). Also described is the implementation of the JH-3 ceramic model and capabilities for modelling projectile dwell phenomena. The Johnson-Holmquist series of ceramic models (JH-1, JH-2, and JH-3) is currently being used in a broader program aimed at computational optimization of composite lightweight armor for Future Combat System vehicles.

Acknowledgments

The author would like to acknowledge Timothy Holmquist, Network Computing Services, Incorporated, for providing helpful discussions during the implementation of the model.

Table of Contents

		<u>Pa</u>	ge
	Acknowledgments		iii
	List of Figures		vii
	List of Tables	•	ix
1.	Introduction		1
2.	Constitutive Model		1
3.	Numerical Implementation		3
3.1	JH-2 Model		4
3.2	JH-3 Model		4
3.3	Impact Onto Ceramic Targets		5
4.	Summary		5
5.	References		17
	Distribution List		19
	Report Documentation Page		39

List of Figures

F	igur	<u>e</u>	Page
	1.	Validation of JH-2 model with analytical expressions, equations (3) and (4).	7
	2.	Excess compression $\mu = \frac{V_0}{V} - 1$ vs. time	. 8
	3.	X-stress vs. time (JH-2 model)	. 8
	4.	Y-stress vs. time (JH-2 model)	. 9
	5.	Z-stress vs. time (JH-2 model)	. 9
	6.	Pressure vs. time (JH-2 model)	10
	7.	X-stress vs. time (JH-3 model)	10
	8.	Y-stress vs. time (JH-3 model)	11
	9.	Z-stress vs. time (JH-3 model)	11
	10.	Pressure vs. time (JH-3 model)	12
	11.	Effective stress vs. pressure for the JH-2 and JH-3 ceramic models	12
	12.	Axial and lateral stress vs. time for 1-D plate impact (JH-2 model)	13
	13.	Axial and lateral stress vs. time for 1-D plate impact (JH-3 model)	13

List of Tables

<u>Table</u>			Page	
1.	Algorithm for the JH-2 ceramic material model		14	
2.	JH-2 model constants for surrogate alumina		15	

1. Introduction

This report describes the implementation of a fully three-dimensional (3-D) rate, pressure and damage-dependent constitutive model for brittle materials such as ceramics into the explicit, Lagrangian finite element code DYNA3D. The DYNA3D source code was obtained from Lawrence Livermore National Laboratories (LLNL) under the auspices of the U.S. Army Research Laboratory (ARL)-LLNL collaborator's agreement. The model, otherwise known as the Johnson-Holmquist II (JH-2) ceramic model, has also been implemented into CTH, EPIC, and LS-DYNA (model 110 in version 960 of the commercial code developed by Livermore Software Technology Corporation [LSTC]), and is used extensively in modelling the brittle response of ceramics in armor applications. This report outlines the important features of the JH-2 material model, the numerical implementation, and validation of the model implementation using the DYNA3D material model driver for constant strain-rate input histories. Despite the fact that the DYNA3D finite element code possesses over 40 constitutive models, few models (e.g., Fahrentold Brittle Damage [model 40]; Defense Threat Reduction Agency [DTRA] Concrete/Geological material [model 45]) are capable of simulating the rate- and pressure-dependent behavior observed in brittle ceramic materials. Hence, the current JH-2 model implementation provides an important phenomenological tool for armor designers involved in simulation-based design of composite integral armors.

2. Constitutive Model

The original Johnson-Holmquist brittle ceramic model (JH-1) is described in Johnson and Holmquist (1992), and represents a phenomenological description of brittle material behavior that includes pressure-dependent strength, strain-rate effects, and damage induced dilatation or bulking effects. In the following year, the model was improved to account for, among other things, the gradual-softening behavior that is observed in ceramics subjected to flyer-plate impact (Johnson and Holmquist 1993). The improved JH-2 model was implemented in both Lagrangian (EPIC) and Eulerian (CTH) hydrocodes where it was used in a number of studies aimed at predicting the depth-of-penetration (DOP) of long rod penetrators. Interestingly, even though the "so-called" improved JH-2 model captures the post-failure behavior of ceramics, it tends to overpredict the DOP (e.g., on the order of 13% [Meyer 1995] in AD995), while the JH-1 model tends to underpredict the DOP by only a few percent. This behavior is attributed to the fact that the JH-2 ceramic model continually degrades the yield strength as damage accumulates 0 < D < 1, whereas strength degrades instantaneously for the JH-1 model only after D = 1. The most recent JH-3 model incorporates most features of the JH-2 model, most notably the power-law relationship between effective stress and pressure, as well as ceramic dwell phenomenology. More recently, researchers using the JH-1 ceramic model in the EPIC code have demonstrated that ceramic dwell can only be simulated using hybrid algorithms that couple Lagrangian and particle methods (e.g., general particle algorithms [GPA]) (Beissel and Johnson 2001). Apparently, tensile waves are artificially introduced into the computations when Lagrangian elements are removed from the calculations during penetration, which can cause premature

material failure. However, the use of hybrid Lagrangian/meshless methods may not be the only means to computationally simulate ceramic dwell phenomena; the development of more robust microphysically based constitutive models for ceramics or novel element erosion algorithms (Simha et al. 2002) might also provide a means for accurately modelling projectile dwell without the need to resort to particle or other meshless methods.

A thorough description of the JH-2 model can be found in Johnson and Holmquist (1993), and only those equations needed for description of the algorithmic implementation are reproduced here. Generally speaking, the normalized equivalent stress can be written as a power-law function of hydrostatic pressure shown in Figure 1, which illustrates continuous curves for the intact strength, σ_i^* , where D=0, damaged strength, σ^* , where 0< D<1, and fully fractured strength, σ_f^* , where D=1, for the ceramic material. The general expression governing these curves is given by

$$\sigma^* = \sigma_i^* - D(\sigma_i^* - \sigma_f^*) \quad , \tag{1}$$

where D is a scalar damage parameter defined over the range $0 \le D \le 1$, and the stresses in equation (1) are made dimensionless by normalizing them to the equivalent stress at the Hugoniot Elastic Limit (HEL) through

$$\sigma^* = \sigma/\sigma_{HEL} \quad . \tag{2}$$

The normalized intact strength σ_i^* is given by

$$\sigma_i^* = A(P^* + T^*)^N (1 + C \ln \dot{\varepsilon}^*) \quad ,$$
 (3)

and the normalized fracture strength σ_f^* is given by

$$\sigma_f^* = B(P^*)^M (1 + C \ln \dot{\varepsilon}^*) \quad , \tag{4}$$

are illustrated in Figure 1 for C=0. Pressure in equations (3) and (4) are made dimensionless by normalizing to the pressure at the HEL through

$$P^* = P/P_{HEL} \quad , \tag{5}$$

and $T^* = T/P_{HEL}$ is the normalized maximum tensile fracture strength. Note that T^* approaches zero as D approaches 1. The additional material constants are A, B, C, M, and N. All damage is assumed to accumulate through incremental plastic $\Delta \varepsilon^p$ deformation of the ceramic using an expression similar to that in the Johnson-Cook fracture model (Johnson and Cook 1985)

$$D = \sum \Delta \varepsilon^p / \varepsilon_f^p \quad , \tag{6}$$

where the rate-independent plastic strain to fracture is given by the power-law expression

$$\varepsilon_f^p = D_1 (P^* + T^*)^{D_2} \quad . \tag{7}$$

Under dynamic loading, the equation-of-state (EOS) for the brittle-material can be defined by a polynomial expression that is written in terms of the excess compression μ (see also Kohn [1969])

$$P = K_1 \mu + K_2 \mu^2 + K_3 \mu^3 + \Delta P \quad , \tag{8}$$

where $\mu = \rho/\rho_0 - 1$, and ρ_0 , ρ are the initial and final densities, respectively. K_1 , K_2 , and K_3 , are constants determined from plate impact or diamond anvil press experiments. A pressure increment ΔP is added when damage begins to accumulate, D > 0, and represents an increase in potential energy as the deviatoric stresses decrease due to material softening. The internal energy U is quadratically related to the equivalent plastic flow stress σ_y

$$U = \sigma_y^2 / 6G \quad , \tag{9}$$

where G is the shear modulus. Hence, the incremental energy loss ΔU can be computed as the difference between successive damage states using

$$\Delta U = U_{D(t)} - U_{D(t+\Delta t)} \quad . \tag{10}$$

If the energy loss ΔU is converted to a potential hydrostatic pressure increase ΔP , an approximate energy conservation relation can be written as

$$\left(\Delta P_{t+\Delta t} - \Delta P_t\right)\mu_t + \left(\Delta P_{t+\Delta t}^2 - \Delta P_t^2\right)/2K_1 = \beta \Delta U \quad , \tag{11}$$

where β governs the fraction of elastic energy converted to potential energy. The implementation utilizes the current excess compression μ_t in contrast to the updated value $\mu_{t+\Delta t}$ used by Johnson and Holmquist (1992). Solving for the updated $\Delta P_{t+\Delta t}$ yields the expression

$$\Delta P_{t+\Delta t} = -K_1 \mu_t + \sqrt{(K_1 \mu_t + \Delta P_t)^2 + 2\beta K_1 \Delta U} \quad . \tag{12}$$

3. Numerical Implementation

A flowchart depicting the numerical implementation algorithm for the equations described in the previous section is outlined in Table 1. Validation of the algorithm and numerical implementation is also demonstrated using the DYNA3D material driver which allows the user to specify an arbitrary strain input history for prediction of the stress state at a material point. The material driver capability present in DYNA3D permits investigation of fundamental material behavior in the absence of inertial or wave propagation effects. The DYNA3D implementation is compared to the results obtained for a "one-element" EPIC model that is constrained laterally, but subjected to a slowly increasing and then decreasing axial force (Johnson and Holmquist 1993). Figure 2 illustrates the DYNA3D uniaxial strain input history representing loading and unloading of the ceramic in the x-coordinate direction

at constant strain rate. The strain history is analytically described by the sawtooth function

$$\mu(t) = 5tH(t)/100 - (t-1)H(t-1)/10, \tag{13}$$

where H(t) is the Heaviside step function.

3.1 JH-2 Model

The material response is illustrated using a series of figures depicting behavior of a surrogate ceramic material subjected to uniaxial strain history. Model constants are given in Johnson and Holmquist (1993) but are reproduced here in Table 2 (case C) for all plots shown in the following sections. Figures 3, 4, and 5 illustrate the stress response of the ceramic in the respective x, y, and z coordinate directions, where x is the loading direction. Figure 6 shows the continuous pressure vs. time behavior. The plot of effective stress vs. pressure (Figure 1) demonstrates validation of the JH-2 model implementation into DYNA3D by comparison with plots in Johnson and Holmquist (1993) (Figure 4 in that reference). Figure 1 also compares the JH-2 model results superimposed upon the analytical expressions given in equations 3 and 4 for intact D=0, and fully fractured D=1 ceramic, respectively. As previously described in Johnson and Holmquist (1993), the material response for this load history is quite complex and can be described by referring to the letters a-f in Figure 1. From the unstressed state, the material begins to load elastically to point a, which intersects the intact strength curve (equation [3]) in Figure 1. The material begins to "soften" as damage accumulates from point a (D = 0) to b (D = 1), where the fracture strength curve (equation [4]) is intersected. The material continues to flow plastically from point b to c along the fracture strength curve and reaches a maximum pressure of 7.25 GPa at point c corresponding to the maximum compression of $\mu = 0.05$. The material then unloads elastically from point c to d where the effective stress is zero. From point d to e the unloading continues and the axial deviatoric stress becomes tensile; the fracture stress is encountered at point e and continues to unload along the fracture strength envelope to point f. In the DYNA3D example, the stresses do not return to zero as in the example provided in Johnson and Holmquist (1993) because the former is under strain control and there are residual stresses induced by yielding, whereas the latter is under stress control.

3.2 JH-3 Model

The JH-3 model was also implemented into DYNA3D since recent work indicates that the defeat mechanism known as projectile dwell can be more accurately simulated using the JH-3 model than the JH-2 model when it is used in conjunction with general particle algorithms (GPA) (Beissel and Johnson 2001). The stress response of the JH-3 model for the surrogate ceramic material is shown in Figures 7, 8, and 9 when subjected to the uniaxial strain history in the x coordinate direction (equation [13]). Figure 10 shows the discontinuous pressure vs. time behavior resulting from the abrupt strength loss in the ceramic during failure. Interestingly, the JH-3 model predicts the presence of jump discontinuities in the x, y, and z stresses during failure that can be contrasted to the continuous stress response predicted

by the JH-2 model (e.g., compare Figures 3, 4, and 5 [JH-2 model] with Figures 7, 8, and 9 [JH-3 model]). In particular, the JH-3 model predicts that a jump decrease should occur in the loading direction in a uniaxially strained ceramic with simultaneous jump increases in the lateral stress components. For the JH-3 model, the effective stress continues to increase from point a to b' as damage accumulates, providing the phenomenological mechanism for dwell; failure in the ceramic occurs instantaneously from point b' to c' (Figure 11). The "kink" in the stress drop from point b' to c' in Figure 11 is due to the ceramic bulking phenomenon that occurs in a single cycle in the algorithm; if bulking is not permitted to occur during failure, i.e., $\beta = 0$, then the effective stress drops straight downward to the fully fractured strength curve. In contrast, the effective stress for the JH-2 model gradually degrades as damage accumulates, from point a to b in Figure 11.

3.3 Impact Onto Ceramic Targets

The behavior of the Johnson-Holmquist ceramic model is further studied using a onedimensional (1-D) simulation whereby a 2-in-thick ceramic target is subjected to impact by a semi-infinite, linear elastic flyer plate travelling at an initial velocity of 2×10^4 in/s. The finite-element model geometry is identical to that reported in previous work on impact into functionally graded elastic media where simulation results compared well to closed-form analytical solutions under transient, uniaxial strain, impact loading (Scheidler and Gazonas 2001). The DYNA3D simulations used 60 uniformly spaced hexahedral elements through the thickness of the ceramic target using the ceramic material constants provided in Table 2 (case C). The density and Young's modulus of the flyer plate was $1.8603 \times 10^{-3} \text{ lb}_f \text{s}^2/\text{in}^4$, and 7.583×10^7 psi, respectively. It is seen that the axial and lateral stresses at the center of the JH-2 ceramic target increase monotonically, level-off, and then decrease during the plate impact event (Figure 12). However, for the JH-3 ceramic model, the axial stress increases monotonically, then instantaneously decreases during failure of the ceramic depicted by the arrow at a in Figure 13. The lateral stresses instantaneously increase during failure of the ceramic as depicted by the arrow at b in Figure 13. Since simulations of plate impact tests on ceramics stressed to failure reveal different behaviors for the JH-2 and JH-3 models, plate impact experiments which simultaneously monitor both axial and lateral stresses would aid in verification of the stress predictions provided by these ceramic material models.

4. Summary

This report has described the implementation of a fully 3-D, rate, pressure and damage-dependent constitutive model for brittle materials such as ceramics into the explicit, Lagrangian finite element code DYNA3D. Both the JH-2 and JH-3 ceramic models have been successfully implemented, and their behaviors are compared using the material driver capability present in DYNA3D that permits investigation of fundamental material behavior in the absence of inertial or wave propagation effects. The DYNA3D implementation compares well with results obtained for a "one-element" EPIC model that is constrained laterally, but sub-

jected to a slowly increasing and then decreasing axial force (Johnson and Holmquist 1993).

The JH-3 model was also implemented into DYNA3D since recent work indicates that the defeat mechanism known as projectile dwell can be more accurately simulated using the JH-3 model than the JH-2 model when it is used in conjunction with GPA's. Projectile dwell is phenomenologically modelled using the JH-3 model by essentially delaying the onset of failure, or the decrease in effective stress, until the ceramic is fully damaged, i.e., D=1. Despite this phenomenological advantage over the JH-2 model, the JH-3 model predicts the presence of jump discontinuities in the x, y, and z stresses during failure, which can be contrasted to the continuous stress response predicted by the JH-2 model. In particular, the JH-3 model predicts that a jump decrease should occur in the loading direction in a uniaxially strained ceramic with simultaneous jump increases in the lateral stress components. This behavior is disturbing in view of the uniaxial strain kinematic boundary conditions. Plate impact tests on ceramics stressed to failure with simultaneous monitoring of the axial and lateral stresses (Bourne et al. 1998) would aid in verification of the stress predictions of the JH-2 and JH-3 material model depicted in this report.

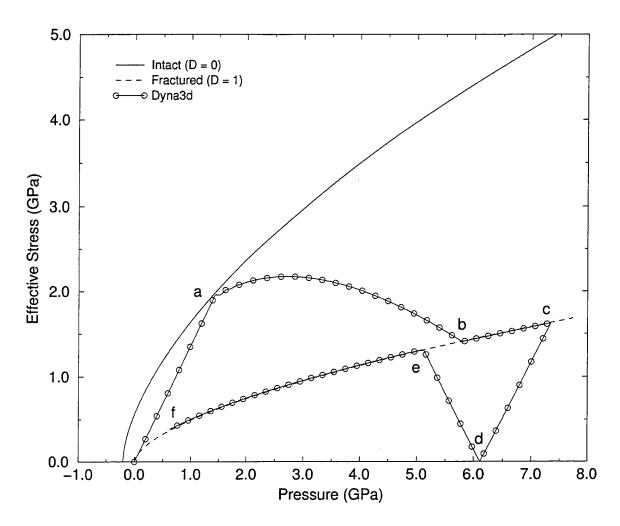


Figure 1. Validation of JH-2 model with analytical expressions, equations (3) and (4).

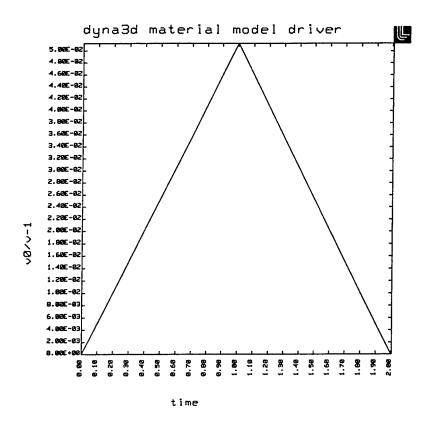


Figure 2. Excess compression $\mu = \frac{V_0}{V} - 1$ vs. time.

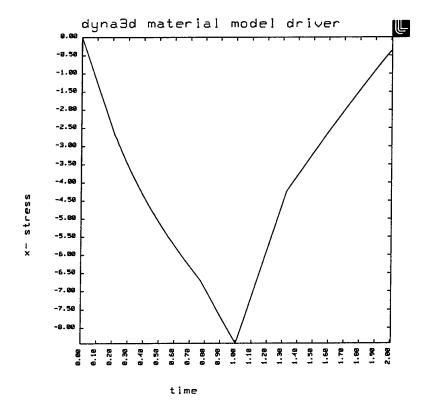


Figure 3. X-stress vs. time (JH-2 model).

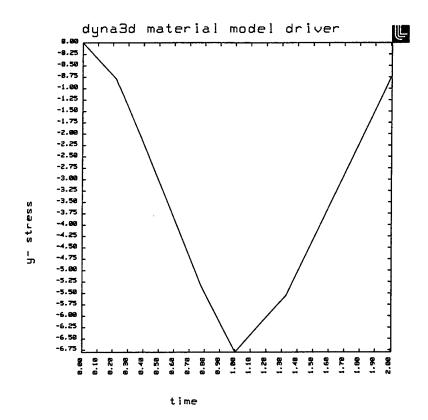


Figure 4. Y-stress vs. time (JH-2 model).

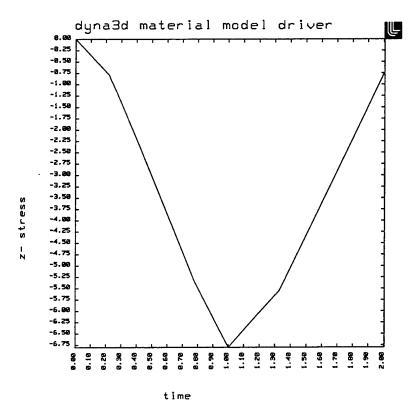


Figure 5. Z-stress vs. time (JH-2 model).

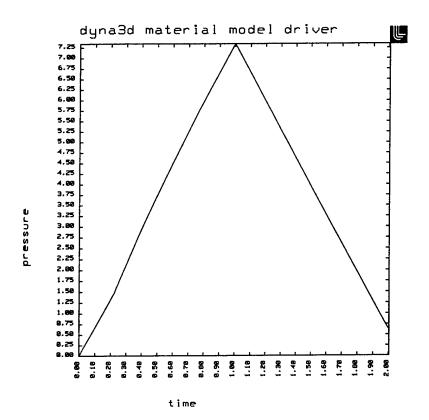


Figure 6. Pressure vs. time (JH-2 model).

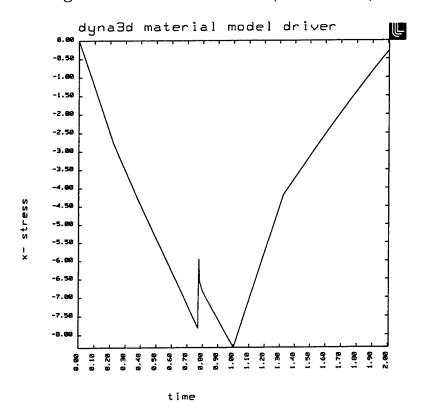


Figure 7. X-stress vs. time (JH-3 model).

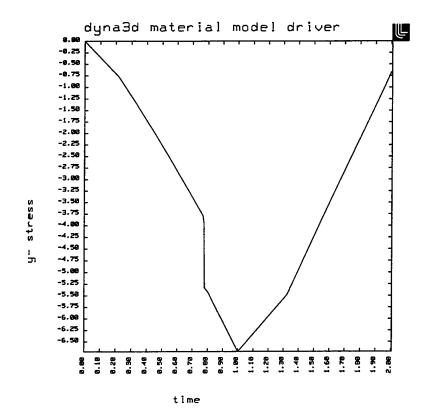


Figure 8. Y-stress vs. time (JH-3 model).

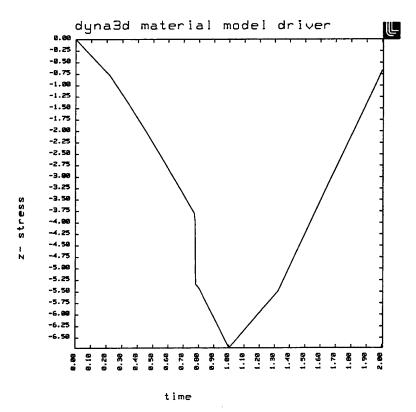


Figure 9. Z-stress vs. time (JH-3 model).

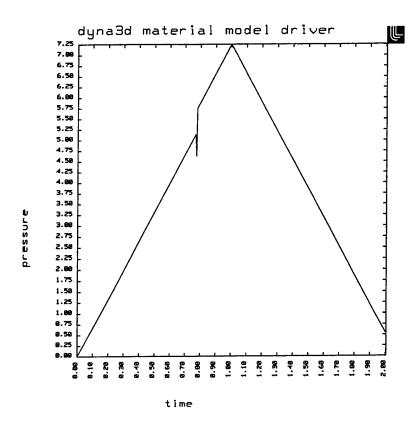


Figure 10. Pressure vs. time (JH-3 model).

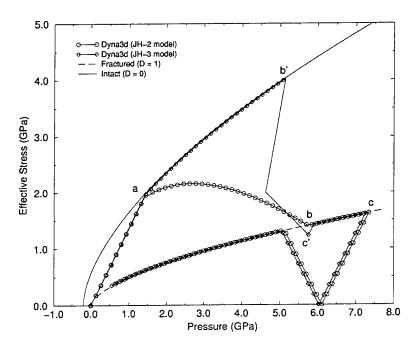


Figure 11. Effective stress vs. pressure for the JH-2 and JH-3 ceramic models.

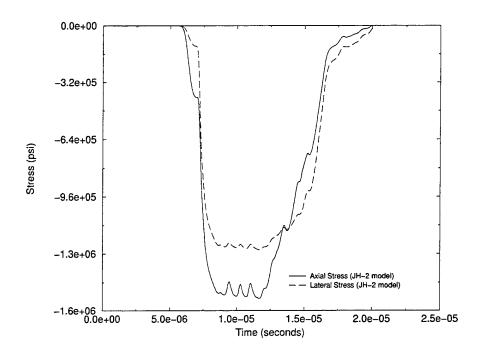


Figure 12. Axial and lateral stress vs. time for 1-D plate impact (JH-2 model).

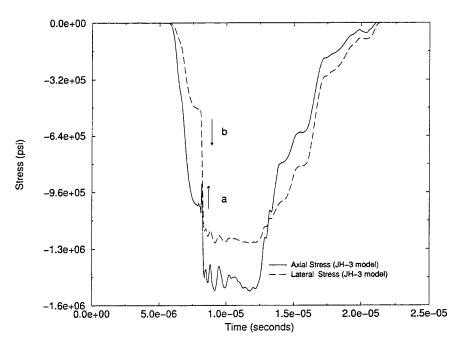


Figure 13. Axial and lateral stress vs. time for 1-D plate impact (JH-3 model).

Table 1. Algorithm for the JH-2 ceramic material model.

1. Calculate trial deviatoric stresses:

$$\sigma_{n+1}^{tr} = \sigma_n + 2\mu \dot{e}\Delta t .$$

2. Calculate total effective strain rate $\dot{\varepsilon}$, and $\dot{\varepsilon}^* = \dot{\varepsilon}/\dot{\varepsilon}_o$:

$$\dot{\varepsilon} = \sqrt{\frac{2}{3}\dot{\varepsilon}_{ij}\dot{\varepsilon}_{ij}} \ .$$

3. Calculate the current yield stress σ^* (equation [1]):

$$\sigma^* = \sigma^*(\dot{\varepsilon}^*, D, P^*, T^*) .$$

4. Calculate the trial effective stress:

$$\bar{\sigma}^{tr} = \sqrt{\frac{3}{2}S^{tr}:S^{tr}} .$$

5. Check for yielding:

$$\bar{\sigma}^{tr} - \sigma^* > 0$$
; continue with step 6.

$$\bar{\sigma}^{tr} - \sigma^* \leq 0$$
; go to step 7.

6. Return stresses radially to yield surface:

$$S_{n+1} = \frac{\sigma^*}{\bar{\sigma}^{tr}} S^{tr} .$$

- 7. Calculate effective plastic strain rate $\dot{\varepsilon}^p$.
- 8. Calculate plastic strain to fracture: ε_f^p (equation [7]) .
- 9. Update damage:

$$D = D + \sum \frac{\dot{\varepsilon}^p \Delta t}{\varepsilon_f^p} \ .$$

- 10. Calculate pressure P and sound speed c.
- 11. Calculate internal energy loss Δu due to damage (equation [10]).
- 12. Calculate pressure increment Δp (equation [12]).
- 13. Update pressure (equation [8]):

$$P = K_1 \mu + K_2 \mu^2 + K_3 \mu^3 + \Delta P$$

14. Reduce deviator stresses by energy ratio:

$$S_{n+1} = S_{n+1} \cdot ratio$$
.

15. Calculate new total stress:

$$\sigma_{n+1} = S_{n+1} + p_{n+1}\delta.$$

16. Go to 1.

Table 2. JH-2 model constants for surrogate alumina.

Table 2. JH-2 model constants for surrogate adminia.					
Parameter	Description	Case A	Case B	Case C	
ρ_o (g/cc)	Density	3.890	3.890	3.890	
G (GPa)	Shear modulus	90.160	90.160	90.160	
HEL (GPa)	Hugoniot elastic limit (HEL)	2.790	2.790	2.790	
P_{HEL} (GPa)	Pressure component at HEL	1.460	1.460	1.460	
σ_{HEL} (GPa)	$\sigma_{HEL} = \frac{3}{2} \left(HEL - P_{HEL} \right)$	1.995	1.995	1.995	
A	Intact strength constant	0.930	0.930	0.930	
N	Intact strength constant	0.600	0.600	0.600	
C	Strain rate constant	0.000	0.000	0.000	
B	Fracture strength constant	0.000	0.000	0.310	
M	Fracture strength constant	0.000	0.000	0.600	
$S_{f\mathrm{max}}^*$	Normalized maximum fracture strength	1.000	1.000	1.000	
$\mid T^* \mid$	Normalized tensile strength	0.137	0.1369	0.137	
K_1 (GPa)	Bulk modulus	130.950	130.950	130.950	
K_2 (GPa)	EOS constant	0.000	0.000	0.000	
K_3 (GPa)	EOS constant	0.000	0.000	0.000	
β	Bulking factor	1.000	1.000	1.000	
D_1	Damage coefficient	0.000	0.005	0.005	
D_2	Damage exponent	1.000	1.000	1.000	

5. References

- Beissel, S., and G. R. Johnson. "Modeling High-Velocity Impact and Explosive Detonation With Finite Elements and Meshless Particles." *Army HPC Research Center Bulletin*, vol. 11, pp. 5–9, 2001.
- Bourne, N., J. Millett, Z. Rosenberg, and N. Murray. "On the Shock Induced Failure of Brittle Solids." *Journal of the Mechanics and Physics of Solids*, vol. 46, pp. 1887–1980, 1998.
- Johnson, G. R., and W. H. Cook. "Fracture Characteristics of Three Metals Subjected to Various Strains, Strain Rates, Temperatures, and Pressures." *Journal of Engineering Fracture Mechanics*, vol. 21, pp. 31–85, 1985.
- Johnson, G. R., and T. J. Holmquist. "A Computational Constitutive Model for Brittle Materials Subjected to Large Strains, High Strain Rates and High Pressures." Shock Wave and High-Strain-Rate Phenomena in Materials, edited by M. A. Meyers, L. E. Murr, and K. P. Staudhammer, New York: Marcel Dekker, 1992.
- Johnson, G. R., and T. J. Holmquist. "An Improved Computational Constitutive Model for Brittle Materials." *High Pressure Science and Technology*, New York: AIP Press, 1993.
- Kohn, B. J. "Compilation of Hugoniot Equations of State." AFWL-TR-69-38, Air Force Weapons Laboratroy, Air Force Systems Command, Kirtland Air Force Base, NM, 1969.
- Meyer, H. W. Jr. "A Comparison of the JH1 and JH2 Ceramic Models in CTH." The Society for Computer Simulation, Proceedings of the 1996 Simulation Multiconference: Military, Government, and Aerospace Simulation, vol. 28, pp. 194–198, edited by M. Chinni, 1995.
- Scheidler, M. J., and G. A. Gazonas. "Analytical and Computational Study of One-Dimensional Impact of Graded Elastic Solids." Proceedings of the 12th APS Topical Conference on Shock Compression of Condensed Matter, Atlanta, GA, June 2001.
- Simha, C. H. M., S. J. Bless, and A. Bedford. "Computational Modeling of the Penetration Response of a High-Purity Ceramic." *International Journal of Impact Engineering*, vol. 27, pp. 65–86, 2002.

NO. OF COPIES ORGANIZATION

- 2 DEFENSE TECHNICAL
 INFORMATION CENTER
 DTIC OCA
 8725 JOHN J KINGMAN RD
 STE 0944
 FT BELVOIR VA 22060-6218
- 1 HQDA DAMO FDT 400 ARMY PENTAGON WASHINGTON DC 20310-0460
- 1 OSD
 OUSD(A&T)/ODDR&E(R)
 DR R J TREW
 3800 DEFENSE PENTAGON
 WASHINGTON DC 20301-3800
- 1 COMMANDING GENERAL US ARMY MATERIEL CMD AMCRDA TF 5001 EISENHOWER AVE ALEXANDRIA VA 22333-0001
- 1 INST FOR ADVNCD TCHNLGY THE UNIV OF TEXAS AT AUSTIN 3925 W BRAKER LN STE 400 AUSTIN TX 78759-5316
- 1 US MILITARY ACADEMY
 MATH SCI CTR EXCELLENCE
 MADN MATH
 THAYER HALL
 WEST POINT NY 10996-1786
- 1 DIRECTOR
 US ARMY RESEARCH LAB
 AMSRL D
 DR D SMITH
 2800 POWDER MILL RD
 ADELPHI MD 20783-1197
- 1 DIRECTOR
 US ARMY RESEARCH LAB
 AMSRL CI AI R
 2800 POWDER MILL RD
 ADELPHI MD 20783-1197

NO. OF COPIES ORGANIZATION

- 3 DIRECTOR
 US ARMY RESEARCH LAB
 AMSRL CI LL
 2800 POWDER MILL RD
 ADELPHI MD 20783-1197
- 3 DIRECTOR
 US ARMY RESEARCH LAB
 AMSRL CI IS T
 2800 POWDER MILL RD
 ADELPHI MD 20783-1197

ABERDEEN PROVING GROUND

2 DIR USARL AMSRL CI LP (BLDG 305)

NO. OF NO. OF COPIES ORGANIZATION COPIES ORGANIZATION 1 **COMMANDER** 1 DIRECTOR US ARMY RESEARCH LAB US ARMY ARDEC AMSRL CP CA AMSTA AR FSE PICATINNY ARSENAL NJ **D SNIDER** 07806-5000 2800 POWDER MILL RD ADELPHI MD 20783-1145 **COMMANDER** US ARMY ARDEC 1 DIRECTOR AMSTA AR TD US ARMY RESEARCH LAB C SPINELLI AMSRL CI IS R PICATINNY ARSENAL NJ 2800 POWDER MILL RD ADELPHI MD 20783-1145 07806-5000 **COMMANDER** 3 DIRECTOR US ARMY RESEARCH LAB US ARMY ARDEC AMSTA AR CCH A AMSRL OP SD TL 2800 POWDER MILL RD W ANDREWS ADELPHI MD 20783-1145 S MUSALLI R CARR DPTY ASST SECY FOR R&T **M LUCIANO** 1 SARD TT E LOGSDEN THE PENTAGON T LOUZEIRO PICATINNY ARSENAL NJ RM 3EA79 07806-5000 WASHINGTON DC 20301-7100 **COMMANDER** 1 **COMMANDER** US ARMY MATERIEL CMD US ARMY ARDEC AMSTA AR CCH P AMXMI INT **JLUTZ** 5001 EISENHOWER AVE PICATINNY ARSENAL NJ ALEXANDRIA VA 22333-0001 07806-5000 **COMMANDER COMMANDER** US ARMY ARDEC US ARMY ARDEC AMSTA AR CC AMSTA AR FSF T **G PAYNE** C LIVECCHIA **J GEHBAUER** PICATINNY ARSENAL NJ C BAULIEU **HOPAT** 07806-5000 PICATINNY ARSENAL NJ **COMMANDER** 07806-5000 US ARMY ARDEC AMSTA ASF 2 **COMMANDER** PICATINNY ARSENAL NJ US ARMY ARDEC 07806-5000 AMSTA AR AE WW E BAKER **COMMANDER** J PEARSON US ARMY ARDEC PICATINNY ARSENAL NJ AMSTA AR QACTC 07806-5000 C PATEL PICATINNY ARSENAL NJ 07806-5000

NO. OF NO. OF COPIES ORGANIZATION COPIES ORGANIZATION COMMANDER US ARMY ARDEC US ARMY ARDEC INTELLIGENCE SPECIALIST AMSTA AR WELF AMSTA AR M M GUERRIERE D DEMELLA PICATINNY ARSENAL NJ PICATINNY ARSENAL NJ 07806-5000 07806-5000 3 COMMANDER 9 **COMMANDER** US ARMY ARDEC US ARMY ARDEC AMSTA AR FSA AMSTA AR CCH B A WARNASH P DONADIA **B MACHAK** F DONLON M CHIEFA **PVALENTI** PICATINNY ARSENAL NJ C KNUTSON 07806-5000 **G EUSTICE** SPATEL 2 **COMMANDER G WAGNECZ** US ARMY ARDEC R SAYER AMSTA AR FSP G F CHANG M SCHIKSNIS PICATINNY ARSENAL NJ D CARLUCCI 07806-5000 PICATINNY ARSENAL NJ 07806-5000 **COMMANDER** US ARMY ARDEC **COMMANDER** AMSTA AR CCL US ARMY ARDEC F PUZYCKI AMSTA AR CCH C R MCHUGH H CHANIN **D CONWAY** S CHICO E JAROSZEWSKI PICATINNY ARSENAL NJ R SCHLENNER 07806-5000 M CLUNE PICATINNY ARSENAL NJ **COMMANDER** 07806-5000 US ARMY ARDEC AMSTA AR OAC T PM ARMS D RIGOGLIOSO SFAE GCSS ARMS PICATINNY ARSENAL NJ BLDG 171 07806-5000 PICATINNY ARSENAL NJ 07806-5000 COMMANDER US ARMY ARDEC PEO FIELD ARTILLERY SYS AMSTA AR WET SFAE FAS PM T SACHAR **H GOLDMAN**

T MCWILLIAMS

07806-5000

PICATINNY ARSENAL NJ

BLDG 172

07806-5000

PICATINNY ARSENAL NJ

NO. OF COPIES	<u>ORGANIZATION</u>	NO. OF COPIES	<u>ORGANIZATION</u>
1	COMMANDER US ARMY ARDEC AMSTA AR WEA J BRESCIA PICATINNY ARSENAL NJ 07806-5000	1	DIRECTOR AIR FORCE RESEARCH LAB MLLMD D MIRACLE 2230 TENTH ST WRIGHT PATTERSON AFB OH 45433-7817
12	PM TMAS SFAE GSSC TMA R MORRIS C KIMKER D GUZIEWICZ E KOPACZ R ROESER	1	OFC OF NAVAL RESEARCH J CHRISTODOULOU ONR CODE 332 800 N QUINCY ST ARLINGTON VA 22217-5600
	R DARCY R KOWALSKI R MCDANOLDS L D ULISSE C ROLLER	1	US ARMY CERL R LAMPO 2902 NEWMARK DR CHAMPAIGN IL 61822
	J MCGREEN B PATTER PICATINNY ARSENAL NJ 07806-5000	1	COMMANDER US ARMY TACOM PM SURVIVABLE SYSTEMS SFAE GCSS W GSI H M RYZYI
1	COMMANDER US ARMY ARDEC PRODUCTION BASE		6501 ELEVEN MILE RD WARREN MI 48397-5000
	MODERN ACTY AMSMC PBM K PICATINNY ARSENAL NJ 07806-5000		COMMANDER US ARMY TACOM CHIEF ABRAMS TESTING SFAE GCSS W AB QT T KRASKIEWICZ
	COMMANDER US ARMY TACOM PM ABRAMS		6501 ELEVEN MILE RD WARREN MI 48397-5000
	SFAE ASM AB 6501 ELEVEN MILE RD WARREN MI 48397-5000		COMMANDER WATERVLIET ARSENAL SMCWV QAE Q B VANINA
	COMMANDER US ARMY TACOM AMSTA SF		BLDG 44 WATERVLIET NY 12189-4050
	WARREN MI 48397-5000	-	ARMOR SCHOOL ATZK TD
	COMMANDER US ARMY TACOM PM BFVS SFAE GCSS W BV 6501 ELEVEN MILE RD WARREN MI 48397-5000		R BAUEN J BERG A POMEY FT KNOX KY 40121

NO. OF NO. OF COPIES ORGANIZATION COPIES ORGANIZATION DIRECTOR HQ IOC TANK AMMUNITION TEAM US ARMY AMCOM SFAE AV RAM TV **AMSIO SMT** D CALDWELL R CRAWFORD BLDG 5300 W HARRIS **ROCK ISLAND IL 61299-6000** REDSTONE ARSENAL AL 35898 **COMMANDER** US ARMY AMCOM NAVAL SURFACE WARFARE CTR 1 AVIATION APPLIED TECH DIR DAHLGREN DIV CODE G06 DAHLGREN VA 22448 J SCHUCK FT EUSTIS VA 23604-5577 US ARMY CORPS OF ENGINEERS 14 **COMMANDER** CERD C **US ARMY TACOM** T LIU AMSTA TR R **CEW ET** R MCCLELLAND T TAN **D THOMAS** 20 MASS AVE NW J BENNETT **WASHINGTON DC 20314** D HANSEN AMSTA JSK US ARMY COLD REGIONS S GOODMAN RSCH & ENGRNG LAB **J FLORENCE** P DUTTA K IYER 72 LYME RD **D TEMPLETON** HANOVER NH 03755 A SCHUMACHER AMSTA TR D USA SBCCOM PM SOLDIER SPT D OSTBERG AMSSB PM RSS A L HINOIOSA **J CONNORS B RAIU** KANSAS ST AMSTA CS SF NATICK MA 01760-5057 H HUTCHINSON F SCHWARZ USA SBCCOM WARREN MI 48397-5000 MATERIAL SCIENCE TEAM AMSSB RSS BENET LABORATORIES 14 I HERBERT AMSTA AR CCB M SENNETT R FISCELLA KANSAS ST M SOJA NATICK MA 01760-5057 E KATHE M SCAVULO OFC OF NAVAL RESEARCH G SPENCER D SIEGEL CODE 351 P WHEELER J KELLY S KRUPSKI 800 N QUINCY ST J VASILAKIS **ARLINGTON VA 22217-5660 G FRIAR** R HASENBEIN NAVAL SURFACE WARFARE CTR 1 AMSTA CCB R **TECH LIBRARY CODE 323** S SOPOK 17320 DAHLGREN RD E HYLAND **DAHLGREN VA 22448 D CRAYON** R DILLON

WATERVLIET NY 12189-4050

NO. OF COPIES	<u>ORGANIZATION</u>	NO. OF COPIES	ORGANIZATION
1	NAVAL SURFACE WARFARE CTR CRANE DIVISION M JOHNSON CODE 20H4 LOUISVILLE KY 40214-5245	8	US ARMY SBCCOM SOLDIER SYSTEMS CENTER BALLISTICS TEAM J WARD W ZUKAS
2	NAVAL SURFACE WARFARE CTR U SORATHIA C WILLIAMS CD 6551 9500 MACARTHUR BLVD WEST BETHESDA MD 20817		P CUNNIFF J SONG MARINE CORPS TEAM J MACKIEWICZ BUS AREA ADVOCACY TEAM W HASKELL
2	COMMANDER NAVAL SURFACE WARFARE CTR CARDEROCK DIVISION R PETERSON CODE 2020 M CRITCHFIELD CODE 1730 BETHESDA MD 20084	9	AMSSB RCP SS W NYKVIST S BEAUDOIN KANSAS ST NATICK MA 01760-5019 US ARMY RESEARCH OFC
	DIRECTOR US ARMY NATIONAL GROUND INTELLIGENCE CTR D LEITER MS 404 M HOLTUS MS 301 M WOLFE MS 307 S MINGLEDORF MS 504 J GASTON MS 301 W GSTATTENBAUER MS 304 R WARNER MS 305 J CRIDER MS 306 220 SEVENTH ST NE CHARLOTTESVILLE VA 22091	9	A CROWSON H EVERETT J PRATER G ANDERSON D STEPP D KISEROW J CHANG B LAMATTINA J LAVERY PO BOX 12211 RESEARCH TRIANGLE PARK NC 27709-2211 NAVAL SURFACE WARFARE CTR
	NAVAL SEA SYSTEMS CMD D LIESE 2531 JEFFERSON DAVIS HWY ARLINGTON VA 22242-5160	-	J FRANCIS CODE G30 D WILSON CODE G32 R D COOPER CODE G32 J FRAYSSE CODE G33 E ROWE CODE G33
	NAVAL SURFACE WARFARE CTR M LACY CODE B02 17320 DAHLGREN RD DAHLGREN VA 22448		T DURAN CODE G33 L DE SIMONE CODE G33 R HUBBARD CODE G33 DAHLGREN VA 22448
]	EXPEDITIONARY WARFARE DIV N85 F SHOUP 2000 NAVY PENTAGON WASHINGTON DC 20350-2000		NAVAL SURFACE WARFARE CTR CARDEROCK DIVISION R CRANE CODE 2802 C WILLIAMS CODE 6553 3A LEGGETT CIR BETHESDA MD 20054-5000

NO. OF COPIES	ORGANIZATION	NO. OF COPIES	ORGANIZATION
1	AFRL MLBC 2941 P ST RM 136 WRIGHT PATTERSON AFB OH 45433-7750 AFRL MLSS	3	DARPA M VANFOSSEN S WAX L CHRISTODOULOU 3701 N FAIRFAX DR ARLINGTON VA 22203-1714
	R THOMSON 2179 12TH ST RM 122 WRIGHT PATTERSON AFB OH 45433-7718	7	DIRECTOR LOS ALAMOS NATIONAL LAB L HULL MS A133 J V REPA MS A133
2	AFRL F ABRAMS J BROWN BLDG 653 2977 P ST STE 6 WRIGHT PATTERSON AFB OH 45433-7739		J WALTER MS C305 E J CHAPYAK MS F664 P HOWE MS P915 J KENNEDY MS P915 F L ADDESSIO T 3 MS 5000 PO BOX 1633 LOS ALAMOS NM 87545
1	WATERWAYS EXPERIMENT D SCOTT 3909 HALLS FERRY RD SC C VICKSBURG MS 39180	1	OAK RIDGE NATIONAL LABORATORY R M DAVIS PO BOX 2008 OAK RIDGE TN 37831-6195
10	DIRECTOR LLNL R CHRISTENSEN S DETERESA F MAGNESS M FINGER MS 313 M MURPHY L 282	1	OAK RIDGE NATIONAL LABORATORY C EBERLE MS 8048 PO BOX 2008 OAK RIDGE TN 37831
	C HOOVER L125 J LIN L125 R E TIPTON L35 D BAUM L35 T MCABEE L35 PO BOX 808	1	OAK RIDGE NATIONAL LABORATORY C D WARREN MS 8039 PO BOX 2008 OAK RIDGE TN 37831
1	LIVERMORE CA 94550 AFRL MLS OL L COULTER 5851 F AVE BLDG 849 RM AD1A HILL AFB UT 84056-5713		DIRECTOR SANDIA NATIONAL LABS APPLIED MECHANICS DEPT MS 9042 J HANDROCK Y R KAN J LAUFFER PO BOX 969
1	OSD JOINT CCD TEST FORCE OSD JCCD R WILLIAMS 3909 HALLS FERRY RD VICKSBURG MS 29180-6199	1	HO BOX 969 LIVERMORE CA 94551-0969 HYDROGEOLOGIC INC SERDP ESTCP SPT OFC S WALSH 1155 HERNDON PKWY STE 900 HERNDON VA 20170

- 30 SANDIA NATIONAL LABS MAIL SERVICES MS 0100 I ANG MS 0310 P YARRINGTON MS 0310 W TEDESCHI MS 0479 B LEVIN MS 0706 A ROBINSON MS 0819 T TRUCANO MS 0819 P TAYLOR MS 0820 R BRANNON MS 0820 M KIPP MS 0820 D CRAWFORD MS 0820 L CHHABILDAS MS 0821 P STANTON MS 0821 I M MCGLAUN MS 0835 ES HERTEL JR MS 0836 L N KMETYK MS 0980 R REEDER MS 0980 J SOUTHWARD MS 0980 R LAFARGE MS 0986 R TACHAN MS 1156 M FURNISH MS 1168 M FORRESTAL MS 1174 W REINHART MS 1181 D HAYES MS 1181 **J ASAY MS 1181** E W REECE MS 1185 DP KELLY MS 1185 CHALL MS 1209 J COREY MS 1217 C HILLS MS 1411 M VIGIL MS 1454 PO BOX 5800 ALBUQUERQUE NM 87185-0100
- 4 NIST
 M VANLANDINGHAM MS 8621
 J CHIN MS 8621
 J MARTIN MS 8621
 D DUTHINH MS 8611
 100 BUREAU DR
 GAITHERSBURG MD 20899
- 3 NASA LANGLEY RSCH CTR AMSRL VS W ELBER MS 266 F BARTLETT JR MS 266 G FARLEY MS 266 HAMPTON VA 23681-0001

- 1 NASA LANGLEY RSCH CTR T GATES MS 188E HAMPTON VA 23661-3400
- 1 FHWA
 E MUNLEY
 6300 GEORGETOWN PIKE
 MCLEAN VA 22101
- 1 USDOT FEDERAL RAILRD M FATEH RDV 31 WASHINGTON DC 20590
- 1 MARINE CORPS
 INTLLGNC ACTVTY
 D KOSITZKE
 3300 RUSSELL RD STE 250
 QUANTICO VA 22134-5011
- 1 DIRECTOR
 NATIONAL GRND INTLLGNC CTR
 IANG TMT
 220 SEVENTH ST NE
 CHARLOTTESVILLE VA
 22902-5396
- 1 DIRECTOR
 DEFENSE INTLLGNC AGNCY
 TA 5
 K CRELLING
 WASHINGTON DC 20310
- 2 PROTECTION MATERIALS INC M MILLER F CRILLEY 14000 NW 58 CT MIAMI LAKES FL 33014
- 2 FOSTER MILLER
 M ROYLANCE
 W ZUKAS
 195 BEAR HILL RD
 WALTHAM MA 02354-1196
- 1 ROM DEVELOPMENT CORP R O MEARA 136 SWINEBURNE ROW BRICK MARKET PLACE NEWPORT RI 02840

NO. OF NO. OF **COPIES ORGANIZATION** COPIES ORGANIZATION 1 O GARA HESS & EISENHARDT 1 SAIC M GILLESPIE M PALMER 1410 SPRING HILL RD STE 400 9113 LESAINT DR FAIRFIELD OH 45014 MS SH4 5 MCLEAN VA 22102 MILLIKEN RSCH CORP H KUHN OFC DEPUTY UNDER SEC DEFNS 1 M MACLEOD **I THOMPSON** PO BOX 1926 1745 JEFFERSON DAVIS HWY **SPARTANBURG SC 29303** CRYSTAL SQ 4 STE 501 **ARLINGTON VA 22202** 1 CONNEAUGHT INDUSTRIES INC ALLIANT TECHSYSTEMS INC **J SANTOS** PO BOX 1425 I CONDON **COVENTRY RI 02816 E LYNAM I GERHARD** ARMTEC DEFENSE PRODUCTS 1 WV01 16 STATE RT 956 S DYER **PO BOX 210** 85 901 AVE 53 ROCKET CENTER WV 26726-0210 PO BOX 848 COACHELLA CA 92236 PROJECTILE TECHNOLOGY INC 1 515 GILES ST PACIFIC NORTHWEST LAB **HAVRE DE GRACE MD 21078** M SMITH G VAN ARSDALE 5 AEROJET GEN CORP R SHIPPELL D PILLASCH PO BOX 999 T COULTER RICHLAND WA 99352 C FLYNN D RUBAREZUL 2 AMOCO PERFORMANCE M GREINER **PRODUCTS** 1100 WEST HOLLYVALE ST M MICHNO JR AZUSA CA 91702-0296 J BANISAUKAS 4500 MCGINNIS FERRY RD 1 HERCULES INC ALPHARETTA GA 30202-3944 HERCULES PLAZA **WILMINGTON DE 19894** 8 ALLIANT TECHSYSTEMS INC **BRIGS COMPANY** C CANDLAND MN11 2830 1 C AAKHUS MN11 2830 **J BACKOFEN** B SEE MN11 2439 2668 PETERBOROUGH ST N VLAHAKUS MN11 2145 HERNDON VA 22071-2443 R DOHRN MN11 2830 S HAGLUND MN11 2439 1 ZERNOW TECHNICAL SERVICES M HISSONG MN11 2830 L ZERNOW D KAMDAR MN11 2830 425 W BONITA AVE STE 208 600 SECOND ST NE SAN DIMAS CA 91773 HOPKINS MN 55343-8367 GENERAL DYNAMICS OTS L WHITMORE 10101 NINTH ST NORTH ST PETERSBURG FL 33702

- 3 GENERAL DYNAMICS OTS
 FLINCHBAUGH DIV
 E STEINER
 B STEWART
 T LYNCH
 PO BOX 127
 RED LION PA 17356
- 1 GKN AEROSPACE D OLDS 15 STERLING DR WALLINGFORD CT 06492
- 5 SIKORSKY AIRCRAFT
 G JACARUSO
 T CARSTENSAN
 B KAY
 S GARBO MS S330A
 J ADELMANN
 6900 MAIN ST
 PO BOX 9729
 STRATFORD CT 06497-9729
- 1 PRATT & WHITNEY
 C WATSON
 400 MAIN ST MS 114 37
 EAST HARTFORD CT 06108
- 1 AEROSPACE CORP G HAWKINS M4 945 2350 E EL SEGUNDO BLVD EL SEGUNDO CA 90245
- 1 UDLP G THOMAS PO BOX 58123 SANTA CLARA CA 95052
- 2 UDLP
 R BARRETT MAIL DROP M53
 V HORVATICH MAIL DROP M53
 328 W BROKAW RD
 SANTA CLARA CA 95052-0359
- 3 UDLP
 GROUND SYSTEMS DIVISION
 M PEDRAZZI MAIL DROP N09
 A LEE MAIL DROP N11
 M MACLEAN MAIL DROP N06
 1205 COLEMAN AVE
 SANTA CLARA CA 95052

- 2 UDLP
 R BRYNSVOLD
 P JANKE MS 170
 4800 EAST RIVER RD
 MINNEAPOLIS MN 55421-1498
- 2 BOEING ROTORCRAFT
 P MINGURT
 P HANDEL
 800 B PUTNAM BLVD
 WALLINGFORD PA 19086
- BOEING
 DOUGLAS PRODUCTS DIV
 L J HART SMITH
 3855 LAKEWOOD BLVD
 D800 0019
 LONG BEACH CA 90846-0001
- 1 LOCKHEED MARTIN
 SKUNK WORKS
 D FORTNEY
 1011 LOCKHEED WAY
 PALMDALE CA 93599-2502
- 1 LOCKHEED MARTIN
 R FIELDS
 1195 IRWIN CT
 WINTER SPRINGS FL 32708
- MATERIALS SCIENCES CORP
 G FLANAGAN
 C F YEN
 A CAIAZZO
 500 OFC CENTER DR STE 250
 FT WASHINGTON PA 19034
- NORTHRUP GRUMMAN CORP ELECTRONIC SENSORS & SYSTEMS DIV E SCHOCH MS V 16 1745A W NURSERY RD LINTHICUM MD 21090
- 1 GDLS DIVISION D BARTLE PO BOX 1901 WARREN MI 48090

NO. OF	ORGANIZATION	NO. OF COPIES	<u>ORGANIZATION</u>
2	GDLS	1	UCLA MANE DEPT ENGR IV
	D REES		H T HAHN
	M PASIK		LOS ANGELES CA 90024-1597
	PO BOX 2074 WARREN MI 48090-2074	4	UNIV OF DAYTON
	VVARREIN WII 40070-2074	•	RESEARCH INST
1	GDLS		R Y KIM
	MUSKEGON OPERATIONS		A K ROY
	W SOMMERS JR		N BRAR A PIEKUTOWSKI
	76 GETTY ST MUSKEGON MI 49442		300 COLLEGE PARK AVE
	MUSREGON MI 49442		DAYTON OH 45469-0168
1	GENERAL DYNAMICS		
	AMPHIBIOUS SYS	1	UMASS LOWELL
	SURVIVABILITY LEAD		PLASTICS DEPT
	G WALKER 991 ANNAPOLIS WAY		N SCHOTT 1 UNIVERSITY AVE
	WOODBRIDGE VA 22191		LOWELL MA 01854
	WOODBIRD GE VII ZEIDE		20112221111 01001
8	INST FOR ADVANCED	1	IIT RESEARCH CENTER
	TECH		D ROSE
	H FAIR		201 MILL ST ROME NY 13440-6916
	I MCNAB P SULLIVAN		KOME NT 13440-0910
	S BLESS	1	GA TECH RSCH INST
	W REINECKE		GA INST OF TCHNLGY
	C PERSAD		P FRIEDERICH
	J CAZAMIAS		ATLANTA GA 30392
	J DAVIS 3925 W BRAKER LN STE 400	1	MICHIGAN ST UNIV
	AUSTIN TX 78759-5316	1	MSM DEPT
	7105111 17(70707 0010		R AVERILL
2	CIVIL ENGR RSCH FOUNDATION		3515 EB
	PRESIDENT		EAST LANSING MI 48824-1226
	H BERNSTEIN	1	UNIV OF WYOMING
	R BELLE 1015 15TH ST NW STE 600	1	D ADAMS
	WASHINGTON DC 20005		PO BOX 3295
	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		LARAMIE WY 82071
1	ARROW TECH ASSO		
	1233 SHELBURNE RD STE D8	2	PENN STATE UNIV
	SOUTH BURLINGTON VT 05403-7700		R MCNITT C BAKIS
	05405-7700		212 EARTH ENGR
1	R EICHELBERGER		SCIENCES BLDG
	CONSULTANT		UNIVERSITY PARK PA 16802
	409 W CATHERINE ST		
	BEL AIR MD 21014-3613	2	PENN STATE UNIV
			R S ENGEL F COSTANZO
			245 HAMMOND BLDG
			UNIVERSITY PARK PA 16801

- 1 TEXAS A&M UNIV D ALLEN AEROSPACE ENGRNG DEPT COLLEGE STATION TX 77843
- 1 NORTHWESTERN UNIV
 DEPT OF MECHANICAL ENGRNG
 H ESPINOSA
 2145 SHERIDAN RD
 EVANSTON IL 60208
- 1 PURDUE UNIV
 SCHOOL OF AERO & ASTRO
 C T SUN
 W LAFAYETTE IN 47907-1282
- 1 STANFORD UNIV
 DEPT OF AERONAUTICS
 & AEROBALLISTICS
 S TSAI
 DURANT BLDG
 STANFORD CA 94305
- 1 UNIV OF MAINE ADV STR & COMP LAB R LOPEZ ANIDO 5793 AEWC BLDG ORONO ME 04469-5793
- 1 JOHNS HOPKINS UNIV APPLIED PHYSICS LAB P WIENHOLD 11100 JOHNS HOPKINS RD LAUREL MD 20723-6099
- 1 JOHNS HOPKINS UNIV CAMCS K RAMESH LATROBE 122 BALTIMORE MD 21218
- 1 UNIV OF DAYTON
 J M WHITNEY
 COLLEGE PARK AVE
 DAYTON OH 45469-0240
- 2 UNIV OF DELAWARE
 DEPT OF MECH ENGRNG
 T W CHOU
 M SANTARE
 126 SPENCER LABORATORY
 NEWARK DE 19716

- 4 UNIV OF DELAWARE
 CTR FOR COMPOSITE MTRLS
 J GILLESPIE
 S YARLAGADDA
 S ADVANI
 D HEIDER
 201 SPENCER LABORATORY
 NEWARK DE 19716
- 1 UIUC
 DEPT OF MATERIALS
 SCIENCE & ENGINEERING
 J ECONOMY
 1304 WEST GREEN ST 115B
 URBANA IL 61801
- 1 NORTH CAROLINA STATE UNIV CIVIL ENGINEERING DEPT W RASDORF PO BOX 7908 RALEIGH NC 27696-7908
- 1 UNIV OF MARYLAND
 DEPT OF AEROSPACE ENGNRNG
 A J VIZZINI
 COLLEGE PARK MD 20742
- 1 DREXEL UNIV
 A S D WANG
 32ND & CHESTNUT ST
 PHILADELPHIA PA 19104
- 3 UNIV OF TEXAS AT AUSTIN
 CTR FOR ELECTROMECHANICS
 J PRICE
 A WALLS
 J KITZMILLER
 10100 BURNET RD
 AUSTIN TX 78758-4497
- 3 VA POLYTECHNICAL
 INST & STATE UNIV
 DEPT OF ESM
 M W HYER
 K REIFSNIDER
 R JONES
 BLACKSBURG VA 24061-0219

NO. OF COPIES	ORGANIZATION	NO. OF COPIES	<u>ORGANIZATION</u>
3	SOUTHWEST RSCH INST ENGR & MATL SCIENCES DIV S A MULLIN J RIEGEL J WALKER 6220 CULEBRA RD PO DRAWER 28510	1	JHU MAT SCI & ENG DEPT M 21 102 MARYLAND HALL 3400 N CHARLES ST BALTIMORE MD 21218-2689
	SAN ANTONIO TX 78228-0510	5	JHU APPLIED PHYSICS LAB
5	JET PROPULSION LAB IMPACT PHYSICS GROUP Z SEKANINA P WEISSMAN B WEST J ZWISSLER M ADAMS 4800 OAK GROVE DR		T R BETZER A R EATON R H KEITH D K PACE R L WEST JOHNS HOPKINS RD LAUREL MD 20723
3	PASADENA CA 91109 CALTECH M ORTIZ MS 105 50 J SHEPHERD MS 105 50	1	SUNY STONY BROOK DEPT APPL MATH & STAT J GLIMM STONY BROOK NY 11794
1	A P INGERSOLL MS 170 25 1201 E CALIFORNIA BLVD PASADENA CA 91125 CALTECH	1	UC BERKELEY MECHANICAL ENG DEPT GRADUATE OFFICE K LI BERKELEY CA 94720
1	G ORTON MS 169 237 4800 OAK GROVE DR PASADENA CA 91007 GEORGIA INST OF TECH	1	UC DAVIS INST THEORETICAL DYNAMICS E G PUCKETT DAVIS CA 95616
1	COMPUTATIONAL MDLNG CTR S ATLURI ATLANTA GA 30332-0356	2	UC SAN DIEGO DEPT APPLIED MECH & ENG SVCS R011
1	GEORGIA INST OF TECH SCHOOL OF MATL SCI & ENG K LOGAN ATLANTA GA 30332-0245	۰	S N NASSER M MEYERS LA JOLLA CA 92093-0411
1	IOWA STATE UNIV DEPT PHYSICS AND ASTRONOMY J ROSE 34 PHYSICS AMES IA 50011		UNIV OF AL HUNTSVILLE AEROPHYSICS RSCH CTR G HOUGH D J LIQUORNIK PO BOX 999 HUNTSVILLE AL 35899
1	LOUISANA STATE UNIV R W COURTER 948 WYLIE DR BATON ROUGE LA 70808		UNIV OF AL HUNTSVILLE MECH ENGNRNG DEPT W P SCHONBERG HUNTSVILLE AL 35899

NO. OF COPIES	<u>ORGANIZATION</u>	NO. OF COPIES	<u>ORGANIZATION</u>
1	UNIV OF PA DEPT PHYSICS & ASTRONOMY P A HEINEY 209 S 33RD ST PHILADELPHIA PA 19104	1	ITT SCIENCES AND SYSTEMS J WILBECK 600 BLVD SOUTH STE 208 HUNTSVILLE AL 35802
1	UNIV OF TX DEPT OF MECH ENG	1	KAMAN SCIENCES CORP D L JONES 2560 HUNTINGTON AVE STE 200
	E P FAHRENTHOLD AUSTIN TX 78712	7	ALEXANDRIA VA 22303 KAMAN SCIENCES CORP
1	VA POLYTECHNIC INST COLLEGE OF ENGINEERING DEPT ENG SCI AND MECH R C BATRA BLACKSBURG VA 24061-0219	,	J ELDER R P HENDERSON D A PYLES F R SAVAGE J A SUMMERS T W MOORE
2	APPLIED RSRCH ASSOC INC D GRADY F MAESTAS STE A220		T YEM 600 BLVD SOUTH STE 208 HUNTSVILLE AL 35802
	4300 SAN MATEO BLVD NE ALBUQUERQUE NM 87110	3	KAMAN SCIENCES CORP S JONES G L PADEREWSKI
1	COMPUTATIONAL MECHANICS CONSULTANTS J A ZUKAS PO BOX 11314 PARTY CORE AND 2020 2014	1	R G PONZINI 1500 GRDN OF THE GODS RD COLORADO SPRINGS CO 80907 DR KENNEDY & ASSOCING
3	BALTIMORE MD 21239-0314 DYNA EAST CORP		D R KENNEDY & ASSOC INC D KENNEDY PO BOX 4003
	P C CHOU R CICCARELLI W FLIS	2	MOUNTAIN VIEW CA 94040 LIVERMORE SOFTWARE
	3620 HORIZON DR KING OF PRUSSIA PA 19406		TECH CORP J O HALLQUIST T SLAVIK
	DYNASEN J CHAREST M CHAREST		2876 WAVERLY WAY LIVERMORE CA 94550
	M LILLY 20 ARNOLD PL GOLETA CA 93117		LOCKHEED MARTIN ELEC & MSLS G W BROOKS 5600 SAND LAKE RD MP 544 ORLANDO FL 32819-8907
	INTERNATIONAL RSRCH ASSOC D L ORPHAL 4450 BLACK AVE PLEASONTON CA 94566	_	LOCKHEED MARTIN MSL & SPACE SANTA CRUZ FACILITY R HOFFMAN EMPIRE GRADE RD
	R JAMESON 624 ROWE DR ABERDEEN MD 21001		SANTA CRUZ CA 95060

NO. OF COPIES	ORGANIZATION	NO. OF COPIES	<u>ORGANIZATION</u>
1	LOCKHEED MARTIN MSL & SPACE M A LEVIN ORG 81 06 BLDG 598 M R MCHENERY T A NGO ORG 81 10 BLDG 157 111 LOCKHEED WAY SUNNYVALE CA 94088	2 5	SOUTHERN RESEARCH INST L A DECKARD D P SEGERS PO BOX 55305 BIRMINGHAM AL 35255-5305 SRI INTERNATIONAL
4	LOCKHEED MARTIN MSL & SPACE J R ANDERSON W C KNUDSON S KUSUMI 0 81 11 BLDG 157 J PHILLIPS 0 54 50 PO BOX 3504 SUNNYVALE CA 94088	J	J D COLTON D CURRAN R KLOOP R L SEAMAN D A SHOCKEY 333 RAVENSWOOD AVE MENLO PARK CA 94025
1	LOCKHEED MARTIN MSL & SPACE W R EBERLE P O BOX 070017 HUNTSVILLE AL 35807	<u>AE</u> 1	US ARMY MATERIEL
2	NETWORK COMPUTING SVC INC T HOLMQUIST G JOHNSON 1200 WASHINGTON AVE S MINNEAPOLIS MN 55415		SYSTEMS ANALYSIS ACTIVITY P DIETZ 392 HOPKINS RD AMXSY TD APG MD 21005-5071
1	ROCKWELL INTERNATIONAL ROCKETDYNE DIV H LEIFER 16557 PARK LN CIRCLE	1	DIRECTOR US ARMY RESEARCH LAB AMSRL OP AP L APG MD 21005-5066
	LOS ANGELES CA 90049	108	DIR USARL AMSRL CI
1	SAIC M W MCKAY 10260 CAMPUS PT DR SAN DIEGO CA 92121		AMSRL CI S A MARK AMSRL CS IO FI M ADAMSON AMSRL SL BA
1	SHOCK TRANSIENTS INC D DAVISON BOX 5357 HOPKINS MN 55343		AMSRL SL BL D BELY R HENRY AMSRL SL BG AMSRL SL I
	SIMULATION & ENG CO INC E I MULLINS S E MULLINS 8840 HWY 20 STE 200 N MADISON AL 35758		AMSRL WM J SMITH AMSRL WM B A HORST AMSRL WM BA D LYON
	J STERNBERG 20 ESSEX LN WOODBURY CT 06798		AMSRL WM BC P PLOSTINS J NEWILL S WILKERSON A ZIELINSKI

NO. OF

COPIES ORGANIZATION

ABERDEEN PROVING GROUND (CONT)

ABERDEEN PROVING GROUND (CONT)

AMSRL WM BD **B FORCH**

R FIFER

R PESCE RODRIGUEZ

B RICE

AMSRL WM BE

C LEVERITT

AMSRL WM BF J LACETERA

AMSRL WM BR

C SHOEMAKER J BORNSTEIN

AMSRL WM M

D VIECHNICKI

G HAGNAUER

J MCCAULEY

AMSRL WM MA

L GHIORSE

S MCKNIGHT

AMSRL WM MB

B FINK

J BENDER

T BOGETTI

R BOSSOLI

L BURTON

K BOYD

S CORNELISON

P DEHMER

R DOOLEY W DRYSDALE

G GAZONAS

S GHIORSE

D GRANVILLE D HOPKINS

C HOPPEL

DHENRY

R KASTE

M KLUSEWITZ

M LEADORE

R LIEB

E RIGAS

JSANDS

D SPAGNUOLO

W SPURGEON

JTZENG

E WETZEL

A FRYDMAN

B POWERS

B CHEESEMAN AMRSL WM MC

J BEATTY

E CHIN

J MONTGOMERY

A WERECZCAK

J LASALVIA

J WELLS

AMSRL WM MD

W ROY

S WALSH

AMSRL WM T

B BURNS

T W WRIGHT

M ZOLTOSKI

AMSRL WM TA

W GILLICH

H MEYER

T HAVEL

I RUNYEON

M BURKINS **E HORWATH**

B GOOCH

W BRUCHEY

M NORMANDIA

AMRSL WM TB

D KOOKER P BAKER

AMSRL WM TC

T W BJERKE

K KIMSEY M LAMPSON

DSCHEFFLER

S SCHRAML

G SILSBY

B SORENSEN

R SUMMERS

R COATES

AMSRL WM TD A DAS GUPTA

T HADUCH

T MOYNIHAN

F GREGORY

M RAFTENBERG

T WEERASOORIYA

D DANDEKAR

A DIETRICH

E J RAPACKI

S SEGLETES

G R PEHRSON

M SCHEIDLER

S SCHOENFELD

AMSRL WM TE
A NIILER
A PRAKASH
J POWELL
AMSRL SS SD
H WALLACE
AMSRL SS SE DS
R REYZER
R ATKINSON

NO. OF COPIES	ORGANIZATION	NO. OF COPIES	ORGANIZATION
1	LTD R MARTIN MERL TAMWORTH RD HERTFORD SG13 7DG UK	1	DYNAMEC RESEARCH AB AKE PERSSON BOX 201 SE 151 23 SODERTALJE SWEDEN
1	SMC SCOTLAND P W LAY DERA ROSYTH ROSYTH ROYAL DOCKYARD DUNFERMLINE FIFE KY 11 2XR UK	1	ISRAEL INST OF TECHNOLOGY S BODNER FACULTY OF MECHANICAL ENGR HAIFA 3200 ISRAEL
1	CIVIL AVIATION ADMINSTRATION T GOTTESMAN PO BOX 8 BEN GURION INTERNL AIRPORT LOD 70150 ISRAEL	1	DSTO WEAPONS SYSTEMS DIVISION N BURMAN RLLWS SALISBURY SOUTH AUSTRALIA 5108 AUSTRALIA
1	AEROSPATIALE S ANDRE A BTE CC RTE MD132 316 ROUTE DE BAYONNE TOULOUSE 31060 FRANCE		DEF RES ESTABLISHMENT VALCARTIER A DUPUIS 2459 BOULEVARD PIE XI NORTH VALCARTIER QUEBEC CANADA PO BOX 8800 COURCELETTE GOA IRO QUEBEC
	DRA FORT HALSTEAD P N JONES SEVEN OAKS KENT TN 147BP UK	1	CANADA INSTITUT FRANCO ALLEMAND DE RECHERCHES DE SAINT LOUIS
-	DEFENSE RESEARCH ESTAB VALCARTIER F LESAGE COURCELETTE QUEBEC COA IRO CANADA		DE M GIRAUD 5 RUE DU GENERAL CASSAGNOU BOITE POSTALE 34 F 68301 SAINT LOUIS CEDEX FRANCE
_	SWISS FEDERAL ARMAMENTS WKS W LANZ ALLMENDSTRASSE 86 3602 THUN SWITZERLAND	_	ECOLE POLYTECH J MANSON DMX LTC CH 1015 LAUSANNE SWITZERLAND

- 1 TNO DEFENSE RESEARCH
 R IJSSELSTEIN
 ACCOUNT DIRECTOR
 R&D ARMEE
 PO BOX 6006
 2600 JA DELFT
 THE NETHERLANDS
- 2 FOA NATL DEFENSE RESEARCH
 ESTAB
 DIR DEPT OF WEAPONS &
 PROTECTION
 B JANZON
 R HOLMLIN
 S 172 90 STOCKHOLM
 SWEDEN
- 2 DEFENSE TECH & PROC AGENCY
 GROUND
 I CREWTHER
 GENERAL HERZOG HAUS
 3602 THUN
 SWITZERLAND
- 1 MINISTRY OF DEFENCE
 RAFAEL
 ARMAMENT DEVELOPMENT
 AUTH
 M MAYSELESS
 PO BOX 2250
 HAIFA 31021
 ISRAEL
- 1 TNO DEFENSE RESEARCH I H PASMAN POSTBUS 6006 2600 JA DELFT THE NETHERLANDS
- 1 B HIRSCH TACHKEMONY ST 6 NETAMUA 42611 ISRAEL
- 1 DEUTSCHE AEROSPACE AG
 DYNAMICS SYSTEMS
 M HELD
 PO BOX 1340
 D 86523 SCHROBENHAUSEN
 GERMANY

- 1 INST FUR
 MATERIALFORSCHUNG II
 C TSAKMAKIS
 POSTFACH 3640
 FORSCHUNGSZENTRUM
 KARLSRUHE
 D 76021 KARLSRUHE
 GERMANY
- 1 NATL TECH UNIV OF ATHENS DEPT OF ENG SCI I VARDOULAKIS ATHENS 15773 GREECE
- 1 UNIV OF PATRAS
 DEPT OF CIVIL ENGINEERING
 D BESKOS
 PATRAS 26500
 GREECE
- 1 ARISTOTLE UNIV
 OF THESSALONIKI
 DEPT OF MECH & MATLS
 E AIFANTIS
 THESSALONIKI 54006
 GREECE

INTENTIONALLY LEFT BLANK.

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188	
gathering and maintaining the data needed, and collection of information, including suggestions	ormation is estimated to average 1 hour per response, completing and reviewing the collection of information for reducing this burden, to Washington Headquarters	 Send comments regarding this burd s Services. Directorate for information 	den estimate d Operations a	or any other aspect of this and Reports, 1215 Jefferson
Davis Highway, Sulte 1204, Arlington, VA 22202- 1. AGENCY USE ONLY (Leave blank	-4302, and to the Office of Management and Budget, Park) 2. REPORT DATE	3. REPORT TYPE AND	DATES C	1. DC 20503. OVERED
100000000000000000000000000000000000000	March 2002	Final, January-Nov	vember	2001
4. TITLE AND SUBTITLE				ING NUMBERS
Implementation of the John DYNA3D	nson-Holmquist II (JH-2) Cor	istitutive Model Into	622618	3.H80
6. AUTHOR(S)				
George A. Gazonas				
7. PERFORMING ORGANIZATION N U.S. Army Research Laborate				ORMING ORGANIZATION RT NUMBER
ATTN: AMSRL-WM-MB	,		ARL-T	TR-2699
Aberdeen Proving Ground, M	4D 21005-5069			
9. SPONSORING/MONITORING AGENCY NAMES(S) AND ADDRESS(ES)			10.SPONSORING/MONITORING AGENCY REPORT NUMBER	
11. SUPPLEMENTARY NOTES				
12a. DISTRIBUTION/AVAILABILITY S			12b. DIS	TRIBUTION CODE
Approved for public release;	distribution is unlimited.			
13. ABSTRACT (Maximum 200 words	(e)			
This report describes the imp model for brittle materials su otherwise known as the Johns LS-DYNA, and is used exten material driver was used to ve T. J. Holmquist. "An Improve Technology, New York: AI capabilities for modelling pro-	plementation of a fully three-dimension of a fully three-dimension of a fully three-dimension of a fully three-dimension of the explicit ason-Holmquist II (JH-2) ceramic asively in modelling the brittle reperty of the model implementation of the explicit of the model implementation of the explicit of the model in a broader program aims of the explicit of the expli	t, Lagrangian finite ele c model, has also been esponse of ceramics in for constant strain-rate Model for Brittle Mate ed is the implementation Johnson-Holmquist ser	implem armor a input hi erials." on of the	ode DYNA3D. The model, nented into CTH, EPIC, and applications. The DYNA3D istories (Johnson, G. R., and High Pressure Science and the JH-3 ceramic model and teramic models (JH-1, JH-2,
14. SUBJECT TERMS				15. NUMBER OF PAGES
constitutive modelling	VA3D, Johnson-Holmquist, dwe	ell, damage, uniaxial	strain,	42 16. PRICE CODE
47 CEOUDITY OF ACCIDIOATION	Las occupity of account a view	La ofounity of Accition	471011	CO. LIMITATION OF ABOTDACT
17. SECURITY CLASSIFICATION OF REPORT	18. SECURITY CLASSIFICATION OF THIS PAGE	19. SECURITY CLASSIFICA OF ABSTRACT		20. LIMITATION OF ABSTRACT
UNCLASSIFIED	UNCLASSIFIED	UNCLASSIFIE	D !	l TT

39

INTENTIONALLY LEFT BLANK.